7. MATRIX FORCE METHOD

7.1 INTRODUCTION

Conventional force method.



Structure as a whole or any substructure



- 1. Equilibrium of forces.
- 2. Displacement compatibility.
- 3. Force-displacement relation.

Matrix Force Method – also called as Flexibility method.

Member forces are treated as the basic unknowns.

Similar to the classical force method, but based on matrix approach.

Based on **finite element** concept

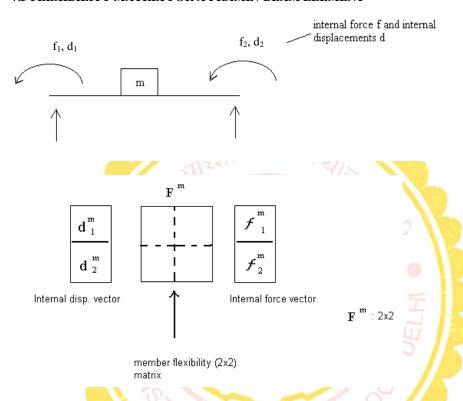


Step-by-step building up of force-displacement relationship using basic elements composing the

7.2 ASSUMPTIONS

- Hooke's law.
- Small deflections.
- Change in length under a deflection \perp to member length = 0.
- Principal of superposition.
- Frames member inextensible.

7.3 FLEXIBILITY MATRIX FOR A FRAME / BEAM ELEMENT



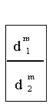
Shears not included since dependent on moments.

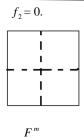
 F_{ij}^{m} = displacement along i^{th} force due to unit force along j^{th} force, all other points being unloaded.

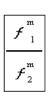
 $f_1 = 1$

Comment [SB1]: This will give the first

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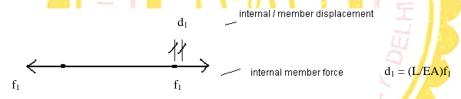
 $\begin{aligned} d_1 &= L/3EI = F_{11}, \\ d_2 &= -L/6EI = F_{21} \\ \text{(derived from slope deflection relations)} \end{aligned}$

Similarly,
$$f_2 = 1 \& f_1 = 0 \Rightarrow 2^{\text{nd}} \text{ column of } \begin{bmatrix} F & m \end{bmatrix}$$

Similarly,
$$F_{12} = -\frac{L}{6EI}$$
 & $F_{22} = \frac{L}{3EI}$

$$F_m = \left(\frac{L}{6EI}\right) \begin{bmatrix} 2 & -1 \\ -1 & 2 \end{bmatrix}$$

7.4 Collective members FLEXIBILITY MATRIX FOR A TRUSS ELEMENT

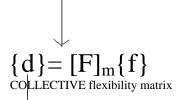


7.5 COLLECTIVE MEMBER FLEXIBILITY MATRIX OF STRUCTURE

Matrix of all internal displacements

Matrix of all internal forces

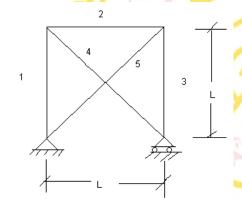
Uncoupled



Internal forces [entire str.]

EXAMPLE:

$$b + r = 5 + 3 = 8$$
 $2j = 2x4 = 8$



$$[F^1] = [F^2] = [F^3] = \left(\frac{L}{EA}\right)$$

$$[\mathbf{F}^4] = [\mathbf{F}^5] = \left\lceil \frac{\sqrt{2}}{EA} L \right\rceil$$

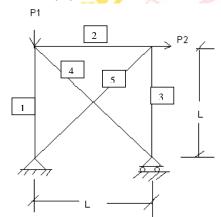
Uncoupled flexibility matrix = [F] =
$$\begin{bmatrix} 1 & 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & \sqrt{2} & 0 \\ 0 & 0 & 0 & 0 & \sqrt{2} \end{bmatrix} \left(\frac{L}{EA} \right)$$
 {d} = [F] {f}

$$\{\mathbf{d}\} = \begin{cases} d_1^1 \\ d_1^2 \\ d_1^3 \\ d_1^4 \\ d_1^5 \end{cases} \qquad \{\mathbf{f}\} = \begin{cases} f_1^1 \\ f_1^2 \\ f_1^3 \\ f_1^4 \\ f_1^5 \end{cases}$$

7.6 TRANSFORMATION OF FORCE (DETERMINATE STRUCTURE)

To find relationship between internal forces and externally applied forces.

Let
$$\{P\} = \begin{cases} P_1 \\ P_2 \\ \dots \\ P_n \end{cases} = \text{Loads applied externally on structure.}$$



How internal forces are related to {P}

$$\mathbf{P} = \begin{cases} P_1 \\ P_2 \end{cases}$$

$$\{f\} \equiv [b] \{P\}$$

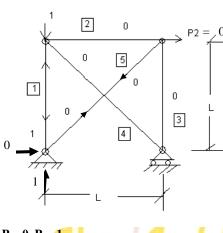
Comment [SB2]: Matrix of all internal forces

 b_{ij} = Internal force f_i caused by unit external force Pj, with all other external forces = 0.

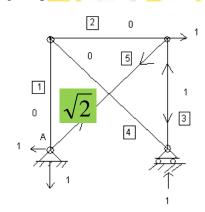
 $P_1 = 1$ $P_2 = 0 \implies f_i$'s will be first column of [b]

 $P_1 = 0$ $P_2 = 1 \Rightarrow f_i$'s will be second column of [b]

$P_1 = 1$ $P_2 = 0$







Apply one by one $\sum F_x = 0$ $\sum M_A = 0$ $\sum F_Y = 0$

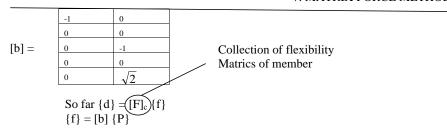
$$\sum F_x = 0$$

$$\sum M_{\perp} = 0$$

$$\sum F_v = 0$$

FTECHNOLOG

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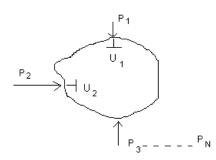
7.7 RELATIONSHIP BETWEEN [P] and [u]

[P] = external loads / forces.

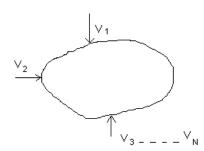
[u] = external displacement.

We will use the principle of virtual work to derive general relation.

Principle of virtual work



Let $U_1, U_2 \dots U_N =$ displacements.



System of virtual forces only.

$\{V\}{<<}\{P\}$

Both {P} and {V} systems are existing simultaneously.

When {P} is applied, {V} rides along the displacements and performs virtual works.

External virtual work By {V} system

Internal virtual works by internal

= forces generated by {V}

{P}

causes deformation {d}

actual

External virtual works = $\{V\}^T \{u\}$

$$\{v\}$$
 internal forces = $[b]$ $\{V\}$ = $\{f\}$ ($\{\delta f\}$ = internal forces)

Therefore internal virtual works= $\{f\}^T \{d\}$

External virtual works = internal virtual works.

$$\{V\}^{T}\{u\} = \{\delta f\}^{T}\{d\}$$
$$= [[b]\{V\}]^{T}\{d\}$$

$$\{u\} = [b]^T \{d\}$$
 $\{d\}$ $\{d\}$ $\{f\}$

$$\{u\} = [b]^T [F]_c \{f\}$$
 $\{f\} - - - - [b] \{P\}$

Therefore, $\{u\} = ([b]^T [F]_c [b]) \{P\}$

 $F_{TS} = \text{Total}$ structural flexibility matrix.

For the truss structure considered earlier,

$$[F_{TS}] = \begin{bmatrix} -1 & 0 & 0 & 0 & 0 \\ 0 & 0 & -1 & 0 & \sqrt{2} \end{bmatrix} \begin{bmatrix} 1 & 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & \sqrt{2} & 0 \\ 0 & 0 & 0 & 0 & \sqrt{2} \end{bmatrix} \begin{pmatrix} L \\ EA \end{pmatrix} \begin{bmatrix} -1 & 0 \\ 0 & 0 \\ 0 & -1 \\ 0 & 0 \\ 0 & \sqrt{2} \end{bmatrix}$$

$$= \begin{bmatrix} -1 & 0 & 0 & 0 & 0 \\ 0 & 0 & -1 & 0 & \sqrt{2} \end{bmatrix} \left(\frac{L}{EA} \right) \begin{bmatrix} -1 & 0 \\ 0 & 0 \\ 0 & -1 \\ 0 & 0 \\ 0 & 2 \end{bmatrix}$$

$$= \left(\frac{L}{EA}\right) \begin{bmatrix} 1 & 0\\ 0 & 1 + 2\sqrt{2} \end{bmatrix}$$

Therefore,
$$\begin{bmatrix} u_1 \\ u_2 \end{bmatrix} = [F]_{TS} \begin{bmatrix} P_1 \\ P_2 \end{bmatrix}$$
$$\begin{cases} u_1 \\ u_2 \end{cases} = \left(\frac{L}{EA}\right) \begin{bmatrix} 1 & 0 \\ 0 & 3.83 \end{bmatrix} \begin{bmatrix} P_1 \\ P_2 \end{bmatrix}$$

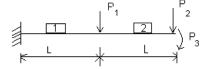
Therefore,
$$u_1 = \frac{P_1 L}{EA}$$
 $u_2 = \frac{3.83 P_2}{EA}$

7.8 PROCEDURE FOR ANALYSIS OF STATICALLY DETERMINATE STRUCTURES

Analysis means to determine the internal forces and the deflections

- 1. Determine {P}
- 2. Define f^m, [F^m], [F]_c
- 3. Form [b]
- 4. internal forces {f}=[b]{P}
- 5. $[F]_{TS} = [b]^T [F]_c [b]$
- 6. $\{u\} = [F]_{TS}\{P\}$

7.9 EXAMPLE 2



$$\{P\} = \begin{bmatrix} P_1 \\ P_2 \\ P_3 \end{bmatrix} \qquad \{u\} = \begin{bmatrix} u_1 \\ u_2 \\ u_3 \end{bmatrix}$$

$$\begin{array}{c|c}
f_2^1 \\
\hline
f_1^1 \\
\hline
f_1^2
\end{array}$$

$$\{f\} = \begin{bmatrix} f_1^1 \\ f_2^1 \\ f_2^1 \\ f_1^2 \\ f_2^2 \end{bmatrix} \qquad \{d\} = \begin{bmatrix} d_1^1 \\ d_2^1 \\ d_2^2 \\ d_2^2 \end{bmatrix}$$
$$[F]_c = \left(\frac{L}{6EI}\right) \begin{bmatrix} 2 & -1 & 0 & 0 \\ -1 & 2 & 0 & 0 \\ 0 & 0 & 2 & -1 \\ 0 & 0 & -1 & 2 \end{bmatrix}$$

$${f} = [b]{P}$$

 ${f} - - - 4x1$
 ${b} - - - 4x3$
 ${P} - - - 3x1$

$$\{f\}$$
---- $4x1$

$$\{P\}$$
----3x1

$P_1 =$	$P_2 =$	P3=
1.0	1.0	1.0

-- All others = 0

$$[b] = \begin{bmatrix} L & 2L & 1 \\ 0 & -L & -1 \\ 0 & L & 1 \\ 0 & 0 & -1 \end{bmatrix}$$

No. of columns = No. of P force.

$$[F]_{TS} = [b]^T [F]_c [b]$$

$$\begin{aligned} & = \left(\frac{L}{6EI}\right) \begin{bmatrix} L & 0 & 0 & 0 \\ 2L & -L & L & 0 \\ 1 & -1 & 1 & -1 \end{bmatrix} \begin{bmatrix} 2 & -1 & 0 & 0 \\ -1 & 2 & 0 & 0 \\ 0 & 0 & 2 & -1 \\ 0 & 0 & -1 & 2 \end{bmatrix} \begin{bmatrix} L & 2L & 1 \\ 0 & -L & -1 \\ 0 & 0 & -1 \end{bmatrix} \\ & = \left(\frac{L}{6EI}\right) \begin{bmatrix} L & 0 & 0 & 0 \\ 2L & -L & L & 0 \\ 1 & -1 & 1 & -1 \end{bmatrix} \begin{bmatrix} 2L & 5L & 3 \\ -L & -4L & -3 \\ 0 & 2L & 3 \\ 0 & -L & -3 \end{bmatrix} \end{aligned}$$

$$= \left(\frac{L}{6EI}\right) \begin{bmatrix} L & 0 & 0 & 0\\ 2L & -L & L & 0\\ 1 & -1 & 1 & -1 \end{bmatrix} \begin{bmatrix} 2L & 5L & 3\\ -L & -4L & -3\\ 0 & 2L & 3\\ 0 & -L & -3 \end{bmatrix}$$

$$= \left(\frac{L}{6EI}\right) \begin{bmatrix} 2L^2 & 5L^2 & 3L \\ 5L^2 & 16L^2 & 12L \\ 3L & 12L & 12 \end{bmatrix}$$
Therefore,
$$\begin{bmatrix} u_1 \\ u_2 \\ u_3 \end{bmatrix} = \left(\frac{L}{6EI}\right) \begin{bmatrix} 2L^2 & 5L^2 & 3L \\ 5L^2 & 16L^2 & 12L \\ 3L & 12L & 12 \end{bmatrix} \begin{bmatrix} P_1 \\ P_2 \\ R_3 \end{bmatrix}$$
Therefore,
$$u_1 = \frac{\left(2L^2P_1 + 5L^2P_2 + 3LP_3\right)}{6EI}$$

$$u_2 = \frac{5L^3P_1 + 16L^3P_2 + 12LP_3}{6EI}$$

$$u_3 = \frac{3L^2P_1 + 12L^2P_2 + 12LP_3}{6EI}$$
If
$$P_2 = P_3 = 0 \implies u_1 = \frac{2P_1L^3}{6EI}$$

$$u_2 = \frac{5P_1L^3}{6EI}$$
,
$$u_3 = \frac{3P_1L^2}{6EI}$$
If
$$P_1 = P_3 = 0$$

$$P_2 = P$$

If the points where displacements are desired are not loaded, we must apply a fictitious load of zero value at those points.

7.10 ANALYSIS OF STATICALLY INDETERMINATE STRUCTURES

INDETERMINATE STRUCTURES: - convert into a "primary" structure by eliminating redundant forces.

Original structure = applied loads + Unknown redundant

$$f_1 = b_{11}P_1 + b_{12}P_2 + \dots + b_{1N}P_n + b_{1(N+1)}x_1 + b_{1(N+2)}x_2 + \dots$$

$$f_2 = \dots + \dots$$

Hence,
$$\{f\} = [b_p \mid b_x] \left[\frac{P}{X} \right]$$

P---- Applied loads

X---- Redundant

 $u_n \rightarrow \mathbf{Unknown}$ Displacements

 $u_x \rightarrow$ **Prescribed** displacement (@ reaction point)

At the points of redundant -

Compatibility condition-- Displacements due to $\{P\}$ + Displacements due to $\{X\}$ = $\{u_x\}$

For a structure on rigid supports, $u_x = 0$

As before-

$$\{u\} = [b]^{T} [F]c[b] \{P\}$$

$$= [b_{p}|b_{x}]^{T} [F]c[b_{p}|b_{x}] \left\{ \frac{P}{X} \right\}$$

$$= \begin{bmatrix} b_{p}^{T} \\ b_{x}^{T} \end{bmatrix} [F_{c}b_{p} \quad F_{c}b_{x}] \left\{ \frac{P}{X} \right\}$$

$$\therefore \begin{bmatrix} \frac{u_{p}}{u_{x}} \end{bmatrix} = \begin{bmatrix} b_{p}^{T}F_{c}b_{p} & b_{p}^{T}F_{c}b_{x} \\ b_{x}^{T}F_{c}b_{p} & b_{x}^{T}F_{c}b_{x} \end{bmatrix} \begin{bmatrix} P \\ X \end{bmatrix}$$

$$\begin{bmatrix} \frac{u_{p}}{u_{x}} \end{bmatrix} = \begin{bmatrix} F_{pp} & F_{px} \\ F_{xp} & F_{xx} \end{bmatrix} \begin{bmatrix} P \\ X \end{bmatrix}$$

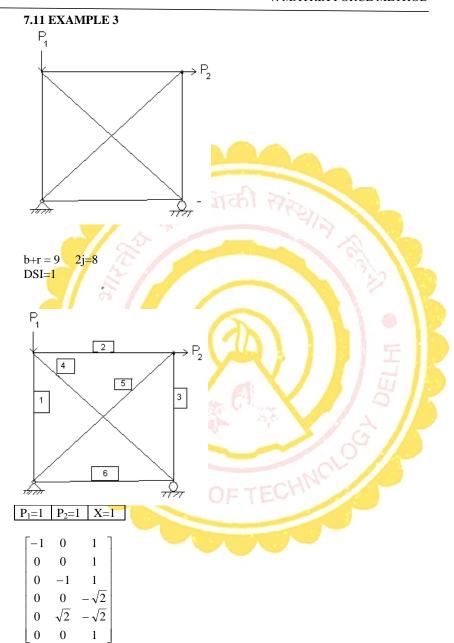
If
$$\{\mathbf{u}_x\} = 0$$
: $0 = [F_{xp}]\{P\} + [F_{xx}]\{x\}$
 $\Rightarrow \{x\} = -[F_{xx}]^{-1}[F_{xp}]\{P\}$
 $\{f\} = [b_p]\{P\} + [b_x]\{X\}$

Unknown Displacements $\{u_p\} = [F_{pp}]\{P\} + [F_{px}]\{X\}$

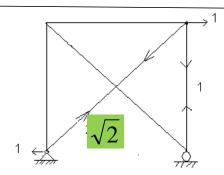
Otherwise ${X} = [F_{xx}]^{-1}({u_x} - [F_{xp}]{P})$

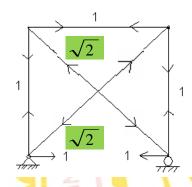
Summarized Procedure for Analysis of Statically Indeterminate Structures

- 1. Define $\{P\}, \{X\}$ (DSI need to be found)
- 2. Define {f},from [F]_c
- 3. $[b] = [b_p|b_x]$ 4. $[F_{pp}] = b_p^T F_c b_c$, $[F_{px}] = b_p^T F_c b_x$, $[F_{xp}] = [F_{px}]^T$, $[F_{xx}] = b_x^T F b_x$ 5. $\{X\} = [F_{xx}]^{-1}(\{u_x\} [F_{xp}]\{P\})$
- 6. $\{f\} = \left[b_p \mid b_x\right] \left\{\frac{P}{X}\right\}$
- 7. $\{u_p\} = [F_{pp}]\{P\} + [F_{px}]\{X\}$



Primary Structure P Make a cut in this member





$$[F]_c = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & \sqrt{2} & 0 & 0 \\ 0 & 0 & 0 & 0 & \sqrt{2} & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{bmatrix} \underbrace{\begin{bmatrix} L \\ EA \end{bmatrix}}$$

$$[F_{pp}] = [b_p]^T [F]_c [b_p]$$

$$[F_{px}] = [b_p]^T [F]_c [b_x]$$

$$[F_{xp}] = [F_{px}]^T = [b_x]^T [F]_c [b_p]$$

$$[F_{xx}] = [b_x]^T [F]_c [b_x]$$

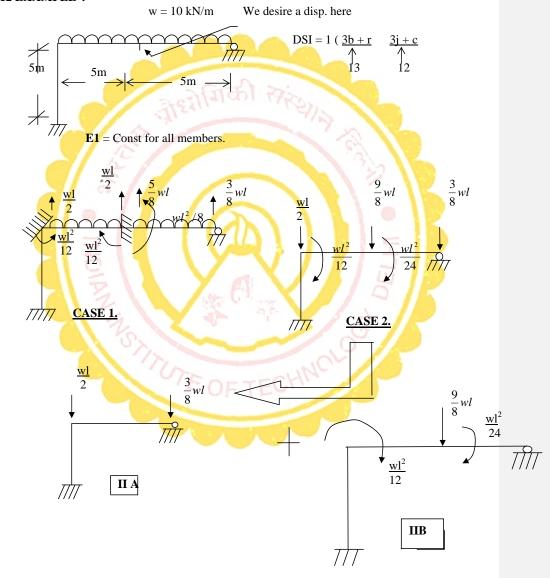
 $\{u_x\} = 0 = \text{Relative Displacements between the cut ends.} \\ \therefore \{X\} = -[F_{xx}]^{-1}[F_{xp}](P)$

$$\therefore \{X\} = -[F_{xx}]^{-1}[F_{xy}]\{P\}$$

Ans.
$$[u_p] = [F_{pp}] \{P\} + [F_{px}] \{X\}$$

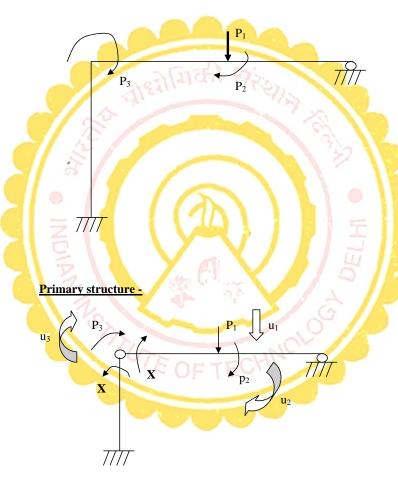
$$\{f\} = [b_p/b_x] \begin{bmatrix} P \\ X \end{bmatrix}$$

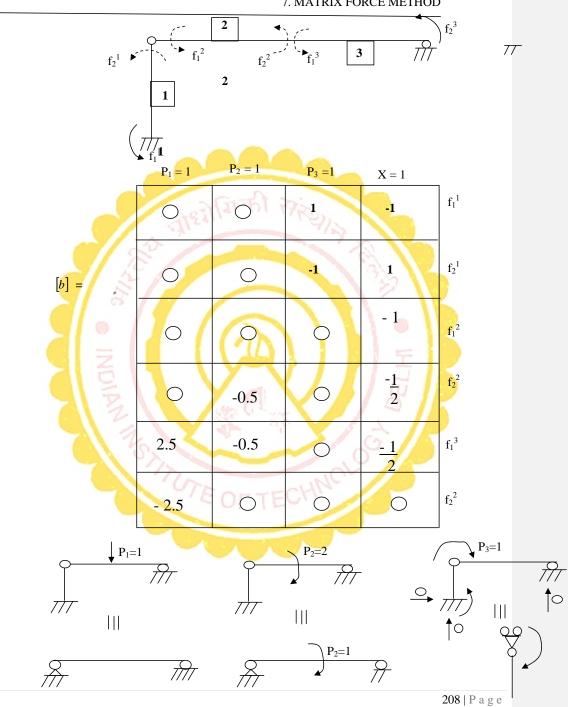
7.12 EXAMPLE 4



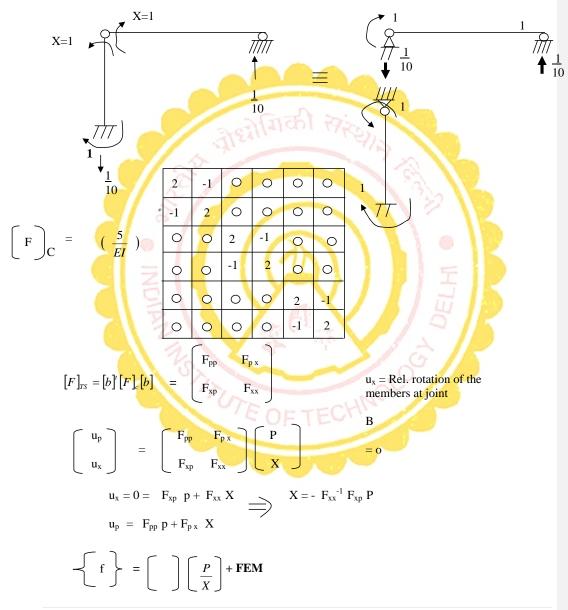
Case II A need not be analyzed.

CASE II B is equivalent to -

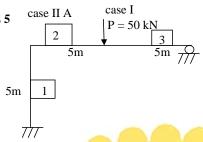




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7.13 **EXAMPLE 5**



<u>3b + r</u>	3j + c
13	12
	DSI = 1



X = Same as Ex. 4

0	-/i
0	1
0	-1
<u>L</u>	$\frac{1}{2}$

 F_c = Same as Ex. 1

$$F_{PP} = b_{p}^{T} F_{c} bp = \frac{L^{3}}{EI} \qquad F_{px} = b_{p}^{T} F_{c} b_{x} = \frac{L^{2}}{EI}$$

$$F_{xp} = b_{x}^{T} F_{c} bp = \frac{3L^{2}}{EI} \qquad F_{xx} = b_{x}^{T} F_{c} b_{x} = \frac{3.5I}{EI}$$

$$X = F_{xx}^{-1} F_{xp} P = 0.857 PL$$

$$U_{p} = F_{pp} P + F_{px} X = \frac{1.428PL^{3}}{EI}$$